



HESSI SPACECRAFT BUS AND INSTRUMENT POWER-ON

HSI_MIT_010E

2000-NOV-15

DAVE CURTIS

DRAFT

As Run on: _____ (Date/Time)

By _____ (Test Conductor)

Test _____

DOCUMENT REVISION RECORD

Rev.	Date	Description of Change
D	2000-11-14	Increase expected current at power-up to account for CPU on; remove software version number recording
E	2000-11-15	Change reference to SAI proc to equivalent HSI_MIT_049 proc

Western Range/NASA Safety: _____ Date _____

Project Manager: _____ Date _____
 Peter Harvey

System Engineer: _____ Date _____
 David Curtis

QA: _____ Date _____
 Ron Jackson

1. INTRODUCTION

1.1 Purpose

This document describes the normal HESSI spacecraft power-up sequence to be run prior to performing functional tests.

2. SETUP

2.1 Test Setup

Verify the test set-up is per the HESSI Spacecraft Power ON/OFF Procedure, HSI_MIT_049, Section 3.0, Spacecraft Power via Test Access Connector (TAC). Verify the following conditions:

- a. Spacecraft (S/C) to Ground Support Equipment (GSE) interface connectors shall be mated at the start of this procedure. Including:
 - TAC-J1 _____ OK
 - BFP-J1 _____ (BFP or Relay Box)
 - UMB-J1 _____ (Yes/No)
 - RF connected to Antenna Hats (RX FWD, TX FWD, RX AFT, TX AFT) _____ (Yes/No)
- b. Transponder enable plug is installed at spacecraft connector FEP J1 (**flight FEP P1 may be used if frangibolt power connectors are demated or solar arrays not installed**).
 - _____ (none, FEP, or TEP)
- c. Integration and Test (I&T) or Flight Battery is installed and mated per HESSI Spacecraft Power ON/OFF Procedure, HSI_MIT_049.
 - _____ (I&T or FLIGHT)
- d. Solar Arrays, Solar Array Simulator or None attached via SW1-4, J1 and J2:
 - _____ (SA, SAS, or None)
- e. List any non-flight configuration of the spacecraft

QA Verify:
TC Verify:

Spacecraft Non-Flight Configurations

1	
2	
3	
4	
5	
6	
7	
8	

3. TEST PROCEDURE

3.1 Spacecraft Power On

- a. Power ON the spacecraft using the Spacecraft Power ON/OFF procedure, HSI_MIT_049, Section 3.0, Spacecraft Power via Test Access Connector (TAC). Record the TAC Voltage and Current.

TAC Voltage		28–36 Vdc
TAC Current		TAC-BATT = 0.5 – 1.4 Adc
Batt Current		

- 1. Verify Mission Mode on the ITOS PACI page is MISSION (not LAUNCH) _____OK
 - If LAUNCH mode,run ITOS script "sc_mm_nom", power off the bus, and start again at step a.
- b. Enable the battery onto the bus:
 - 1. Set the TAC voltage to match the BATT VOLT value on the ITOS PACI telemetry page. _____Volts
 - 2. Enable the battery onto the bus by clicking on the Battery Relay button on the PC _____OK
 - 3. Increase the TAC voltage by 0.2V steps while monitoring the BATT CURRENT on the ITOS PACI page. Increase TAC voltage until BATT CURRENT reads +1Amp (+0.2Amp with I&T battery). _____OK
- c. Enable power to all of the ACS components using the following sequence:
 - 2. Display the “FSW Menu” ITOS telemetry display by typing “page fswmenu,1,1” at the ITOS STOL command prompt.
 - 2. In the FSW Menu telemetry display click on the “PCB Interface” button to display the PCB status ITOS telemetry page.
 - 3. In the “PCB Interface” telemetry display click on the NEB1 “ON” button to command the PCB NEB1 bus on, and verify that the status indicator for NEB1 displays “ON.” _____OK
 - 4. In the “PCB Interface” telemetry display click on the FSS “ON” button to command the PCB to turn on power to the fine sun sensor, and verify that the status indicator for the FSS displays “ON.” _____OK
 - 5. In the “PCB Interface” telemetry display click on the Torque Rod XZ “ON” button to command the PCB to turn on power to the ADB X and Z1

- torque rod drivers, and verify that the status indicator for the Torque Rod XZ displays “ON.” _____OK
- 6. In the “PCB Interface” telemetry display click on the Torque Rod YZ “ON” button to command the PCB to turn on power to the ADB Y and Z2 torque rod drivers, and verify that the status indicator for the Torque Rod YZ displays “ON.” _____OK
- 7. In the “PCB Interface” telemetry display click on the Magnetometer “ON” button to command the PCB to turn on power to the Magnetometer, and verify that the status indicator for the Magnetometer displays “ON.”
- 8. In the FSW Menu telemetry display click on the “ACS Subsystem” button to display the ACS status ITOS telemetry page.
- 9. In the ACS telemetry display verify that the magnetometer magnetic field values are updating. _____OK
- 10. If possible, verify torque rods are active ("singing") _____OK
- 11. Turn off acquisition sequence: "/acssetmode idle" _____OK
- 12. Record the TAC voltage and current:

TAC Voltage		28–36 Vdc
TAC Current		TAC-BATT = 1.0 – 1.5 Adc
Batt Current		

- d. This section powers on the batter pressure monitors
 - a. In the “PCB Interface” telemetry display click on the NEB2 “ON” button to command the PCB NEB2 bus on, and verify that the status indicator for NEB2 displays “ON.” _____OK
 - b. In the "PCB Interface" telemetry display click on the SPARE 1 "ON" button to command on the battery pressure monitor, and verify the status indicator for SPARE 1 displays "ON" _____OK
 - c. Record the Battery Pressure telemetry on the "PACI" page

Batt Press 1		3000-5000
Batt Press 2		3000-5000

- d. This section powers on the SSR and prepares it for use

1. Start the procedure SSR_ON. Verify SSR status is ON on the "PCB Interface" page _____ OK
~~— if ETU SSR is installed, start procedure SSRSIMMAKEPAT —~~ _____ ~~OK~~
2. Record TAC voltage and current:

TAC Voltage		28–36 Vdc
TAC Current		TAC-BATT = 1.3 – 1.9 Adc
Batt Current		

3.2 Power on the Instrument

- a. This section powers on the IDPU. Instrument displays should be on a separate ITOS computer if possible.
 - 1. Display the "IGSE Menu" ITOS telemetry display by typing "page igse_pages" at the ITOS STOL command prompt.
 - 2. In the IGSE menu click on the "SOH Executive" button and the "IDPU Voltages" button
 - 3. Start the "SC_IDPUON" procedure. In the "PCB Interface" page verify that the status indicator for the IDPU displays "ON." Verify IDPU current is less than 0.7A _____ OK
 - 4. Verify that the IDPU_MODE on the "SOH Executive" displays "Normal" _____ OK
 - 5. Record any errors on "SOH Executive" page (SPECTEMP error is normal if detectors are warm)

Error Count: _____ Errors Code: _____

- 6. Start the "SC_IDPU_SPWRON" procedure. In the "PCB Interface" page verify that the status indicator for the IDPU+28V displays "ON." Verify IDPU LD CUR on "PACI" page is less than 0.1A _____ OK
- 7. Start the "SC_CPCON" procedure. In the "PCB Interface" page verify that the status indicator for the CRYO displays "ON." Verify CRYO Current is less than 0.1A _____ OK
- 8. On the "PCB" page verify no OC trips. _____ OK
- 9. Record TAC voltage and current:

TAC Voltage		28–36 Vdc
TAC Current		TAC-BATT = 1.8 – 2.2 Adc
Batt Current		

- b. Turn on the Imager
 - 1. In the IGSE menu click on the "SOH ADP" button and the "SOH Imager Voltages" button
 - 2. Start the IMGR_ON procedure. Verify that 5 IADP_PWR fields on the SOH ADP page are all "ON". _____ OK
 - 3. Send /IDPUCLEAR to clear errors. Verify that the IDPU_ERRCODE field on the SOH Executive page reads "NONE" _____ OK
 - 4. Send /IADPMODE VALUE=15 to clear ADP errors. Verify that the 4 ADP_STATUS fields on the SOH ADP page read "OK" _____ OK

5. Load the default parameter file: start "v4s0r0_003". Verify all 4 IADP_STATUS values on the SOH ADP page are "OK" _____ OK
6. Send /IADPRUN. Verify all 4 IADP_STATUS values on the SOH ADP page are "OK" _____ OK
7. Record TAC voltage and current:

TAC Voltage		28–36 Vdc
TAC Current		TAC-BATT =
Batt Current		2.0 – 2.6 Adc

c. Turn on the Spectrometer Interface

1. In the IGSE menu click on the "SOH Spec Power" button
2. Display the "IDIB" ITOS page
3. Start the IDIB_ON procedure. Verify that the voltages (+12, +5, -5, -12) for all 9 DIBs are nominal (+/-0.3V) _____ OK
4. Start the IDIB_TM_ON procedure _____ OK
5. Record TAC voltage and current:

TAC Voltage		28–36 Vdc
TAC Current		TAC-BATT =
Batt Current		2.9 – 3.5 Adc

d. Power on the Particle Detector and PMT

1. In the IGSE menu click on the "SOH Particle Detector", "SOH PMT", and "SOH MEM DUMP" buttons
2. Enable the particle detector high voltage. "/IDPUARM PDHV". Verify that the PDHV value on the SOH Particle Detector page reads "ENABLED" _____ OK
3. Turn on the particle detector using the IPD_ON procedure. Verify that the IPDHVDAC value on the SOH Particle Detector page starts to increment once a second. _____ OK
4. Enable PMT VC1 telemetry: On the Downlink ITOS page, select "Bypass Transmitter", then "Stored SOH On", then "IDPU Diagnostics ON" _____ OK

5. Enable the PMT High Voltage: "/IDPUARM PMT". Verify that the PMT HV field on the SOH PMT page indicates ENABLED. _____ OK
6. Turn on the high voltage 28V source supply: start "IHV_ON". Verify that the IDPU_28HV value on the SOH Volages page reads 28V+/-0.5V _____ OK
7. Turn on the PMT: Start "IPMT_ON". Verify that IPMT_HVDAC on the SOH PMT page starts to increment once a second _____ OK
8. Start PMT telemetry: "/IDPUDUMPTABL PMTVARS". _____ OK
9. Wait for IPDHVDAC and IPMT_HVDAC to reach their final values. (IPDHVDAC = 122, IPMT_HVDAC = 130) _____ OK
10. Record bus voltage and current:

TAC Voltage		28-36 Vdc
TAC Current		TAC-BATT = 3.0 - 3.5 Adc
Batt Current		

- e. Increase the TAC current limit on the PC GSE to 6Amps _____ OK
- f. Increase the TAC voltage on the PC GSE in 0.1V steps until the Batt Current is about 0.8A (0.2A if I&T battery), but not greater than 36V. _____ OK
- g. Record Battery Temperature on PACI page. If greater than 25C notify engineer immediately _____ OK
- h. Organize screens so that all data can be seen. Take page snaps of all screens (Instrument and Spacecraft). Be sure that a system clock is on each screen _____ OK

3.3 Collect Trending Data

- a. Start the "Sprt_trnd" ITOS Procedure to collect trending data _____ OK

Completed Date/Time: _____